SAFETY RECOMMENDATIONS SUBMITTED DURING FISCAL YEAR 1964—Continued

Reference No.	Recommendation subject	Submitted	Page No.
4–RO–63 4–RE–64	Amendment of WB Form 652–16A Fleet assembly inspection on all DC-8 stabi- lizer drive assemblies to assure proper as- sembly.	To USWB May 6, 1964 To FAA May 6, 1964	293 295
4-RE-65	FAA change procedures to prevent STC with back-dated effective date; FAA change procedure to prevent installation of in- compatible STC's; change in CAR to prevent certifying aircraft having dangerous loading characteristics.	To FAA May 8, 1964	301
	Inspection and protection of landing gear	To FAA May 13, 1964	309
	Relocation of sensing equipment at Hilo,	To USWB May 22, 1964	312
1–RE–68	FAA consider modification of B-720-B hori-	To FAA May 27, 1964	314
	Recurrent structural inspections of Mooney		320
1–RE–70	1,000-hour retirement life on idler pulley, Hughes 269A/B, be reduced.	do	325
1–R0–71	Replacement of control valve on crew oxygen	To FAA June 5, 1964	329
1-RE-72	bottles. - AD be issued for immediate inspection of	To FAA June 12, 1964	332
1–R0–73	S-10K Fafnir bearings. - Change In wording of par. 265.12 of Air Traffic Control Procedures Manual re special VFR operations.	To FAA June 16, 1964	338
1-R0-74	- Equal application of VFR landing standards to all carriers.	To FAA June 18, 1964	342
-RE-75	Periodic inspection on crankshaft of Piper	To FAA June 19, 1964	344
-R0-76	Twin Comanche. Maintenance of Navaids in condition existing	To FAA June 22, 1964	349
-R 0- 77	at time of accident. - Briefing on location and operation of life rafts	do	352
	and preservers. Inclusion of appropriate provisions of pt. 42 in air tax operators' operations specifications.		356
RE80	Heim rod end failures on Cessna 310 airplanes Landing gear failures, Cessna 120/140 aircraft FAA review quality of work performed by certified repair stations.	do	359 363 368

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Recommendation		Page	
No.	Date	Recommendation	No.
63-1	lulu 2 1002	To FAA: Compliance with service bulletin A-Ty 79-15 at an early date be made	
03-1	July 3, 1302	mandatory to guarantee an ample warning margin in operating engines. (B-96)	
63-2	July 20, 1962	To FAA: Advising FAA of the engine overhaul irregularities found during our investi-	
		gation and recommending that the overhaul practices and procedures of the Pacific	
00.0	44	Airmotive be reviewed by FAA. (B-96)	
63-3	do	To FAA: FAA conduct survey to determine number and type of old emergency escape slides in service and to require removal of such slides from service within a	
		reasonable length of time. (B-96)	
63-4	July 21, 1962	To FAA: Recommending correction of discrepancies in flight information publications	
		and charts. (B-93)	. 1
63–5	July 24, 1962	To FAA: Advising them that the detailed examination of the landing gear fitting dis-	
C2 C	Aug. 10, 1962	closed that the source of the fracture was fatigue	1
03-0	Aug. 10, 1302	WAL procedures and practices with respect to EGT indicating system checks	1
63-7	do	Reduce current overhaul period at Avalon Air Transport from 1,500 to 600 hours;	
		conduct field check of blade slippage within retaining clamps; reevaluate clamp	
		design for retaining and indexing blade in subject model of Hartzell propeller	. 2
63-8	Aug. 14, 1962	Call attention of the Administrator to need for design changes in landing gear door and hinge beyond the manufacturer's change to steel replacement hinge	2
63_9	Aug. 22, 1962	Calling to attention of Administrator failure of section of anti-ice air duct in fuselage	
00 0	Aug. 22, 1002	near aft baggage compartment which was weakened by severe intergranular corro-	
		sion caused by improper pickling	2
63–10	Aug. 23, 1962	1-time inspection be made of all Cessna CH-1C tube braze assemblies and if FAA	
62 11	Aug. 24, 1962	approves new design AD be issued requiring new assembly on all CH-10's	3
03-11	Aug. 24, 1902	Consideration be given to establish shorter periods between inspections and between parts replacements than previously established for such components as landing	
		gears and flaps	3